



Vertical Flight Society
2021-2022 Design-Build-Vertical Flight Competition
Request for Proposal
Updated: March, 1, 2022

Website: www.vtol.org/FLY

Questions: FLY@hq.vtol.org [Changed 3/1/2022]

Executive Summary:

The Vertical Flight Society (VFS) invites student teams to participate in the 2nd Annual VFS Design-Build-Vertical Flight (DBVF) Student Competition. The fly-off portion of the competition is scheduled to take place at the U.S. Army Combat Capabilities Development Command (CCDC) Army Research Laboratory (ARL) Robotics Research Collaboration Campus (R2C2) at [Graces Quarters](#) located in Middle River, Maryland, on **June 1 – 3, 2022**. This remote-control electric vertical takeoff and landing (eVTOL) competition seeks to encourage interest in unmanned aircraft technology and small air vehicle design and fabrication.

With respect to the COVID pandemic, VFS and ARL will continue to monitor any changes to ensure an in-person competition is a safe and viable option. If the competition fly-off does not take place, then the scoring will take into account the deliverables accomplished prior to the competition fly-off.

Team eligibility rules are as follows:

- All participants must be full-time university students, and must have a faculty advisor. Students may be at the undergraduate or graduate level, but they must be currently enrolled during the competition semester.
- Team rosters for the competition flyoff will be limited to a total five (5) students, including any faculty advisors.
- **Teams can have any number of additional student participants (beyond the 5 stated above), which is solely up to the university's discretion.** Teams can also adjust their team rosters as appropriate throughout the competition; with the exception of the students listed to attend the flyoff once the team roster has been submitted to ARL (see below for more details).
 - A total of five (5) team members will be allowed at the competition fly-off, including any faculty advisors. Attendance of at least one (1) Faculty Advisor to the fly-off is recommended but not mandatory. *Note: VFS and ARL reserve the right to reduce or increase the maximum number of team members allowed at the fly-off.*
 - Selected teams will fill out their final team roster of no more than five (5) attendees and submit directly to ARL by **March 28, 2022**, the list will be used to clear students for entrance onto the fly-off location at ARL.
 - Foreign national (non-US Citizen) team members are permitted to attend the fly-off, but additional information will be required by ARL from before they will be cleared to attend the competition. ***Because ARL is a government facility, it is imperative non-US citizen information is fully completed by the deadline date of March 28, 2022 in order to clear the individual to attend the competition. Additionally, some citizens from certain countries may not be eligible to attend in person.***
- The total number of competing teams at the fly-off will be limited to no more than eight (8). If more than eight teams submit LOIs, the top eight (8) teams will be selected based on their Letter of Submission information.
- Each team member, including faculty advisors, must be a current member of VFS (student membership is US\$25/year) at the time of submitting the Preliminary Design Report — find VFS membership information at vtol.org/membership.
- Each team's pilot must also hold an [FAA Part 107 Drone Certification](#). The name of the pilot must be identified in the attendee list submitted directly to ARL on March 28 and must be fully certified before the beginning of the fly-off competition on June 1.

- If there is more than one team per school, VFS reserves the right to limit the number of teams based on review of the initial proposals. In the event that two or more separate teams from the same university apply and must be down-selected, the teams will be given the option to combine into a single team entry that meets all requirements.

The aircraft requirements are detailed in the section(s) below, but note that the aircraft is restricted to a maximum take-off weight (MTOW) of no more than 15 lbs (6.8 kg). The vehicle, along with any power supply and payload, may never exceed 15 lbs. Aircraft size and weight will factor into the competition scoring as described in the scoring section.

Awards

Competition awards will include the following.

- Overall competition winners will be awarded a plaque (1st place) and certificates (2nd and 3rd place) and the following monetary amounts:
 - **1st – \$2000**
 - **2nd – \$1000**
 - **3rd – \$500**
- Monetary prizes will also be awarded following the review and scoring of the Preliminary Design Reports (PDR) for the 1st through 3rd best scores as follows:
 - **1st – \$750**
 - **2nd – \$500**
 - **3rd – \$250**

1.0: Timeline for Deliverables

- Letter of Intent to Submit: **Sept. 1–Oct. 15, 2021**
 - Teams will be notified of their status in early November
- Preliminary Design Report: **Updated to Dec. 20, 2021**
 - Teams will be notified of results in mid to late January 2022
- Team Component List and Network Diagram **to ARL: March 1, 2022**
- Team Fly-off Participant List **to ARL: Updated to March 1, 2022**
- UAS Pilot Certifications Training & Experience **to ARL: due April 25, 2022**
 - Must include Pilot FAA Part 107 Drone Certification Information
- Final Technical Report: **May 2, 2022**
- Competition Fly-off Dates: **June 1-3, 2022**
 - Team Check-in/Pilot Certification: **June 1, 2022**
 - Competition Days: **June 2-3, 2022**

1.1: Letter of Intent (LOI) to Submit

The VFS Design-Build-Vertical Flight Competition proposal submission window will be open from **Sept. 1 to Oct. 15, 2021**; LOIs should be emailed to FLY@HQ.vtol.org.

The LOI submission is limited to 4-pages (including the cover page but excluding the Letter of Support)) and must be written by the students. The LOI should include the following sections:

- Cover page including university and team name
- Team Introduction
- Organization
 - Team roster to include name, email address and class standing
 - Teams can have any number of student participants working on the different phases of the competition. [Note, however, due to ARL occupancy restrictions, only five (5) team members — including any faculty — will be allowed at the competition fly-off (no exceptions). Attendance of at least one (1) faculty advisor at the fly-off is recommended but not mandatory. Refer to Eligibility rules on Page 2]
 - At least one (1) team lead/captain — and no more than two (2) co-leads/co-captains — must be identified. The team captain(s) will be the main contact for all communications related to the competition.
 - Tentative project schedule.
- Overview of Technical Approach
 - Briefly summarize any initial conceptual designs that the team is considering.
- Letters of Support: Please include one (1) letter of support from a Faculty Advisor — letters of support do not count against the 4-page limit.
 - Include any secured or planned team funding, either through the university or outside sponsors.

Teams will be notified of the status of their application to the competition by early November.

1.2: Preliminary Design Report (PDR)

The PDR is limited to 10 pages and is due **Monday, Dec. 20, 2021** at 11:59 PM Eastern Standard Time (UTC-5) and emailed to FLY@HQ.vtol.org. Reminder: team members must be members of VFS at the time of the PDR submission. See vtol.org/membership

All information should be within the 10-page limit, including any reference materials. The details of the report must include (but are not limited to) the list below. The order of information is meant to be a template for how the document is structured — it's preferred that it be in this order for judging criteria; however, it is not mandatory. Scoring criteria for the PDR is in Section 3.1.

- Summary
- Design Trade Studies
- Technical Innovations
- Design Definition
 - Can include a Design Trade Studies section if needed.
- Fabrication Methods
- Schedule

1.3: Final Technical Report (FTR)

The FTR is limited to 15 pages and is due **Monday, April 25, 2022 at 11:59 PM EST (UTC-4); it must be emailed to FLY@HQ.vtol.org.**

All information should be within the 15-page limit, including any referenced materials. The details of this report must include (but are not limited to) the list below. The order of information is meant to be a template for how the document is structured — it's preferred that it be in this order for judging criteria; however, it is not mandatory. Scoring criteria for the FTR is in Section 3.2.

Teams are permitted to re-use relevant portions from their PDR submission (such as Trade Studies) in their FTR. However, teams should consider any judging feedback received from the PDR so that they can improve their FTR submission.

- Executive Summary
- Management Summary
- Design Trade Studies
- Technical Innovations
- Design Definition
- Drawing Package
- Fabrication Methods
- Test Plan
- Flight Test Results

2.0: The Competition

The goal of the final fly-off is to prove the capability of each team's unmanned aircraft system (UAS) as a subscale demonstrator for an advanced air mobility (AAM) aircraft. The courses, course rules and scoring have been determined to test the performance of each aircraft at the subscale level. Final competition scoring will be determined by four parts: the fly-off, the team presentations, and the two team reports, of which the fly-off will carry the most weight in final scoring. The fly-off categories of focus are:

- Range
- Agility and Speed
- Payload Capacity: the payload will be part of the flight performance course (refer to Section 2.3.1)

- The payload quantity cannot be modified mid-attempt. Whatever weight the team chooses must be used for the entire course attempt. Multiple weights can be used to add up to a combined payload weight.
- Payloads must be the commercially available SoftGrip hand weights. These weights can be found online and range in sizes from 0.5–10 lb (0.2–4.5 kg). The hand weights cannot be modified and will be inspected and weighed at the competition. Teams must supply their own SoftGrip payloads.
- The payload can be secured internally, or externally with a safety wire running through the metal grommet securing it to the airframe. The securing of the payload must pass the safety inspection at the competition.
- The payload and attachment are described more in section 2.3.1.

Components, other than batteries, cannot be changed out on the aircraft between phases of the competition. Any component that is used on the aircraft for a single course or aspect of the competition must be on the aircraft for all parts of the competition. Components may be switched 1-for-1 to replace a failed component between flight attempts.

Varying battery packs will be allowed between course attempts, including a 1-for-1 swap or a change in battery capacity that still follows the RFP rules. Any change in battery capacity must not result in the total vehicle weight exceeding the 15 lb limit.

The aircraft configuration is also not allowed to be manually changed for the different challenges. Mechanical systems that actuate components mid-attempt, however, will be allowed. Each team's pilot will be required to demonstrate controlled operation of any such mechanical systems during the pilot-in-command certification portion of the safety checks. For example, a retractable gear, tilting rotors, or a tilting wing would be allowed.

2.1: Judging

A panel of judges will be appointed for the scoring of each aspect of the competition, including the technical reports and team presentations.

The safety tests and course scoring will be carried out by ARL, VFS organizers, and other applicable judges. The majority of points for the fly-off portion of the competition will be awarded based on quantitative metrics, thus removing the potential for bias. The safety tests will be on a pass/fail basis.

2.2: Safety Checks

A series of pilot and safety checkpoints along with a UAS component checklist will be required in order for teams to participate in the final fly-off, which will be conducted by ARL. These tests are put in place to ensure a safe, reliable, properly functioning aircraft to minimize risk during the fly-off. The safety checks will involve a bench check and a hover test.

2.2.1: Bench Check

The bench check will involve each team giving the judge an explanation of how the aircraft works, safety considerations, and operating procedures in front of the aircraft, prior to a thorough inspection. This will

provide an opportunity for judges to ensure the safety of the aircraft, provide an opportunity for any required modifications before moving on, or disqualify an aircraft deemed unsafe or non-compliant.

If an aircraft does not meet all of the safety requirements, the team can apply modifications and ask for an additional bench check. If a team still does not meet the safety requirements, the team will only be allowed to perform the presentation, not the flight missions. For the autonomous mission (as described in Section 2.3.2), the team must prove the remote-kill functionality of the vehicle's power system while operating in autonomy mode during this check.

The judges will be looking for the following:

- The remote controller must demonstrate a lost-link power-cut functionality
- Proper operation of the shunt plug and verification of the correct location
- Battery type verification
- Weighing of the aircraft
- Measurement of the aircraft's maximum dimension
- Checking for suitable integrity and rigidity of the aircraft
- Demonstrating remote-kill functionality in autonomy mode

2.2.2: Hover Test

ARL will run a pilot-in-command certification procedure to certify each team's pilot for operations on the ARL grounds. The pilot certification checklist consists of a straightforward demonstration of basic remote piloting skills and safety. The safety check is not meant to test a pilot's skill, but rather the general ability of the pilot to control their aircraft in routine flight. The specifics of the safety check will be provided to the selected teams. Team pilots will also be required, as part of this check, to demonstrate safe and controlled operation of any mechanical systems used to actuate vehicle components mid-flight, as mentioned in Section 2.0.

The hover test will require each team to power on their aircraft, hover at a specified height, display each directional control input, and make a controlled landing. The aircraft must remain aloft under pilot control for at least 60 seconds before being set back down.

2.3: The Fly-Off

- Team pilots must have proof of their FAA Part 107 Drone Certification (US based teams) or equivalent certifying authority (non-US based teams) in order to compete. *Refer to Section 7.0 Pilot Requirements*
- Participating team's UAS Component List must also meet ARL requirements. *Refer to Section 8.0 ARL Component List Requirements for more detailed information.*
- Teams that successfully complete the safety checks will be permitted to participate in the fly-off. This will consist of two courses — the flight performance course to test aircraft performance, and the autonomy course to test the aircraft's autonomous flight capability.
- The designated test site, located at ARL's R2C2 is shown below in Figure 1. It depicts a 600-ft (183-m)-diameter gravel pad and the location of multiple CONEX shipping containers that comprise a Military Operations in Urban Terrain (MOUT) environment.
- The MOUT environment will be present within the test area and must be accounted for during course runs. A majority of the MOUT building structures are made up of 40' x 9.5' x 8' (LxHxW) shipping containers. These will not be used for the competition.

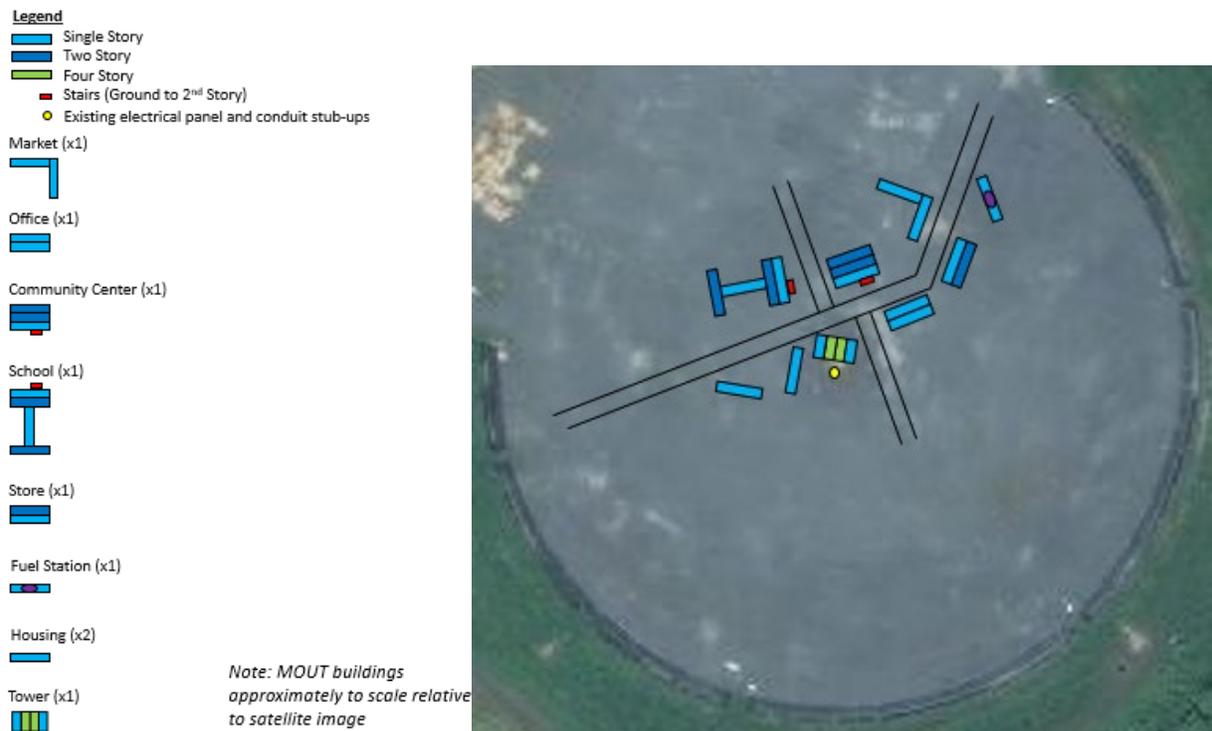


Figure 1: R2C2 Pad Layout with Shipping Containers

2.3.1: Flight Performance Course

This course tests the vehicle’s performance. A known challenge for electric aircraft is the low specific density in current battery technology. This means that to meet functional ranges, the aircraft must operate as efficiently as possible during cruise flight to achieve longer ranges and be marketable for mission success.

- The vehicle will be loaded with at least a 2 lb SoftGrip weight that is provided by the team and is installed before powering on the vehicle, see Figure 2.
- Teams have the opportunity to use a heavier payload as they see fit, subject to the above restrictions and those in Section 2.0.
- The payload fraction will be scored as shown in Table 6.
- The vehicle with any SoftGrip payload will be weighed prior to the flight. The combined weight must be less than 15 lbs.
- The SoftGrip weights can be found and purchased online (e.g. Amazon.com or Walmart.com).
- The payload must be secured to the aircraft, either internally or externally with a safety wire running through the grommet that connects to the airframe.

The course directions are specified below:

- Power on the aircraft (with the chosen payload already installed)
 - The pilot will immediately activate the remote-kill function if either the pilot or the course judge deems it necessary.
- Pilot and navigator enter pilot, navigator and judge box
- Lift off vertically to a height of approximately 5 ft (cleared visually by the course judge)

- Fly around the course as shown in Figure 3
 - Come to a full VTOL landing inside each landing zone (30' diameter circle)
 - The navigator should be equipped with a pair of binoculars or similar, to help the pilot land the vehicle within each landing zone
 - Within each landing zone, it is expected that the aircraft will land vertically and take-off vertically. The aircraft's rotors may remain spinning during the touchdown, but the aircraft must be in contact with the ground and not translating for the range official to waive a flag signaling a satisfactory touchdown. A rolling touch and go is not permitted.
 - A VTOL touch and go is required. As soon as the range official signifies a complete touchdown by waiving their flag, the pilot can take-off and proceed through the course. There is no minimum touchdown time requirement.
 - A range official will be stationed near each landing zone to waive a flag signaling a stationary landing has been satisfied
 - Range officials will be in communication with the course judge via walkie-talkie, and can communicate to the pilot through the course judge if the vehicle has touched down short or long of the landing zone
- The number of completed laps will be recorded for the lap score portion; no partial credit will be given for uncompleted laps.
- A maximum time limit of 10 minutes will be imposed on the course.
- When the aircraft is either low on power, or has surpassed the 10-minute course limit, it must enter the start/end VTOL zone above 5 ft, and descend vertically to a controlled landing
 - Each pilot must make the 'safe-pilot' determination as to when they must return to the start/end VTOL zone.
 - If the vehicle is unable to safely return and land at the start/end VTOL zone (i.e. running out of battery mid-flight), a penalty deduction of 3 complete laps will be incurred.
- Power down the aircraft



Figure 2: SoftGrip 2 lb. Payload (1.0" H x 7.0" L x 2.0" W)

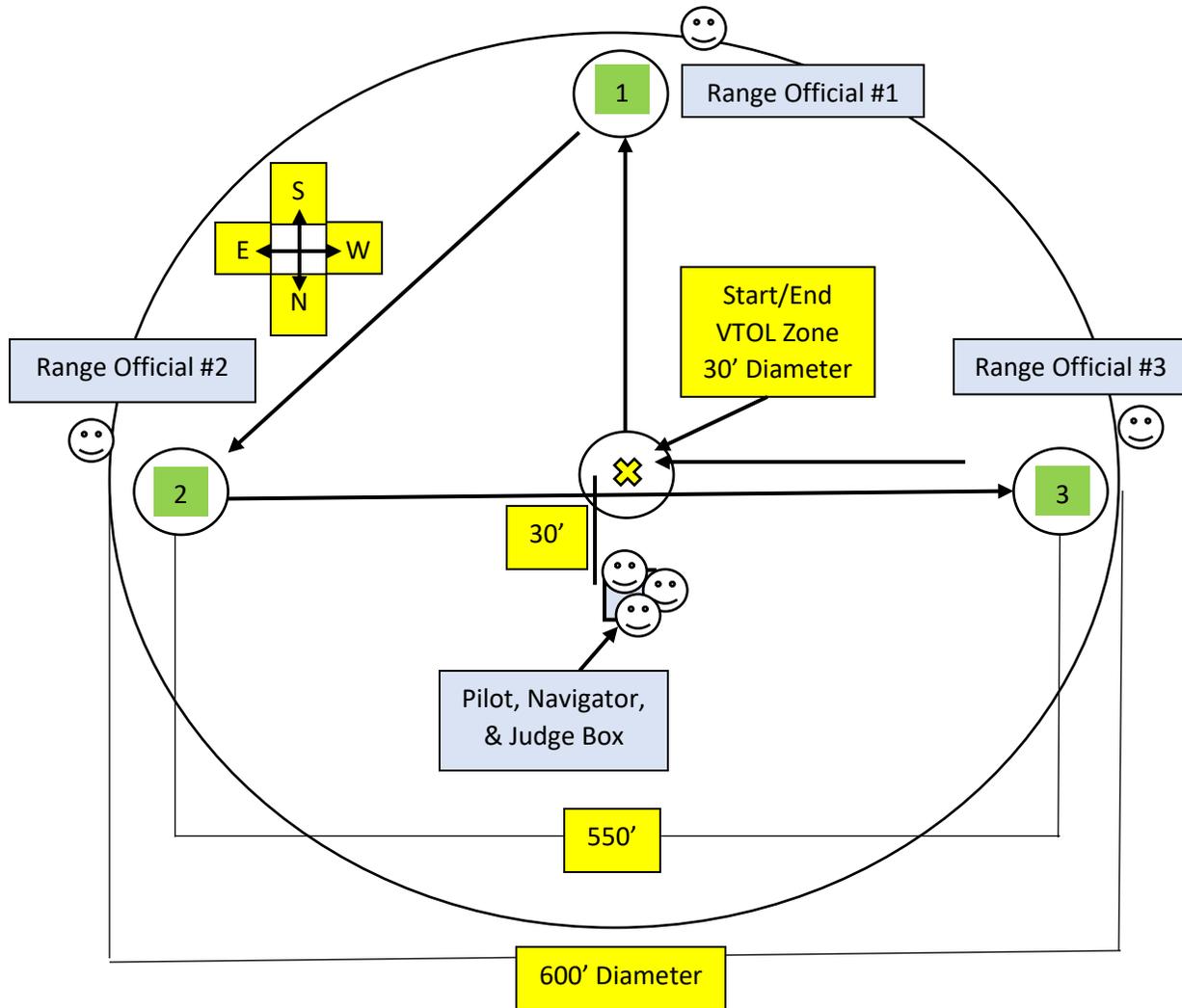


Figure 3: Flight Performance Course Layout

2.3.2: Autonomy Course

The autonomy course requires the same VTOL landing zones from the flight performance course to be programmed via GPS coordinates into the aircraft's control system that the vehicle will then autonomously fly. The course scoring is shown in Table 7.

- The waypoints must be followed in the same order as seen in Figure 3.
- The precise GPS coordinates will be supplied later. Teams attempting the autonomy course are required to prove the remote-kill functionality of the vehicle's power system while operating in autonomy mode during the bench check.
- A "how-to" guide that is specific to autonomy is also being published on the DBVF webpage with recommended equipment and helpful tips (see Section 9 of this document).

The course directions are specified below:

- Power on the aircraft
 - The pilot will immediately activate the remote-kill function if either the pilot or the course judge deems it necessary.

- Pilot enters pilot zone and commands a start of the vehicle's autonomous flight
- Aircraft must climb vertically to a height of approximately 5 ft above the ground, cleared visually by the course judge
- Fly around the course as shown in Figure 3
 - Land and take-off in the general location of each numbered point
- After departing the third waypoint, the aircraft must enter the VTOL zone above 5 ft., and descend vertically to a controlled landing
- Power down the aircraft

2.4: Presentations

Final presentations given by each team will contain an overview of the content from the Final Technical Report. This will involve a deep look into the design drivers based on the initial trade studies, operating procedures, manufacturing methods, and overall project expenses. This material may be presented by no less than 2 people, and no more than 5 people. Each presenter must be a student member of the team. The presentations will be open to all competition participants, with a private Q&A session with the judges following each presentation.

The team should bring their presentation on a USB flash drive. The maximum time allowed for the presentation is 10 minutes. There is a 5-minute question period from the audience and another 5-minute private Q&A session from the judges after each presentation.

The presentation must be in English and should include, but is not limited to:

- Team member introductions
- Trade studies and analysis leading to the selection of the conceptual design used
 - Plots, videos, or images of any quantitative methods, including theoretical design and performance calculations, finite element analysis (FEA), or computational fluid dynamics (CFD) simulations completed
- Final design and fabrication
- Ground and flight testing
 - Videos or images of testing are encouraged

The presentations are graded according to the rubric in section 3.3.

3.0: Scoring

Scoring considerations will include the summation of the following:

- Preliminary Design Report (PDR); 100 Points Possible
 - Scoring carried out by the judges in accordance with the rubric in section 3.1 and will be returned to teams mid-January 2022
- Final Technical Report (FTR); 100 Points Possible
 - Scoring carried out by the judges in accordance with the rubric in section 3.2 and will be returned to teams by mid-May 2022
- Team Presentations; 100 Points Possible
 - Scoring carried out by the judges in accordance with the rubric in section 3.3
- Aircraft Mission Performance; 300 Points Possible
 - Scoring will be carried out by the VFS organizers in accordance with the rubrics in section 3.4

3.1: Preliminary Design Report (PDR) Scoring

Table 1: Preliminary Design Report Scoring Criteria	
Section	Requirement
Summary (5 points)	Overall description of team organization (leadership, sub teams and responsibilities)
	Brief description of problem to be solved (purpose of design)
	Summarizes main point from subsequent sections
Design Trade Studies (15 points)	Mission requirements decomposed into subsystem requirements
	Sensitivity study of design parameters discussed, with design drivers detailed
	Review of configurations considered
	Describe selection process, reasoning and results (e.g. configuration, motors, props, etc.)
Technical Innovations (15 points)	Detail any unique design approaches used by the team (i.e. aircraft configuration)
	Detail any unique technologies or manufacturing techniques being used by the team
	Describe "mission model" used for predicting system performance
	Mission model description includes equations, sources of inputs, assumptions made, and uncertainties
Design Definition (25 points)	Overview of chosen design configuration
	All key subsystems presented (airframe, propulsion system, electronics)
	Document dimensional parameters of design
	Document predicted mission performance
	Provide estimate of aircraft lift and drag and method of prediction
	Provide estimate of static and dynamic stability and method of prediction
	Drawing of aircraft: front view
	Drawing of aircraft: side view
Drawing of aircraft: top view	
Fabrication Methods (15 points)	Manufacturing processes investigated, discussed and compared
	Discussion on how investigated materials and methods were down-selected
	Final manufacturing process presented in detail
Schedule (5 Points)	Schedule includes key actions through report deliverable and presentation
	Schedule includes detail on design, prototype, and testing phases (especially high-risk items)
	Schedule includes detail for subcomponent design
Presentation and Organization (20 points)	Proper grammar, spelling and formatting
	Figures & texts taken from published works are referenced
	Reference list at the end of the document in numerical order as cited in the text
	Logical progression of report; easy to read with headings, etc.
Total (100 points)	

Monetary prizes will also be awarded following the review and scoring of the preliminary design reports for the 1st through 3rd best scores as follows:

- 1st – \$750
- 2nd – \$500
- 3rd – \$250

3.2: Final Technical Report (FTR) Scoring

Table 2: Final Technical Report Scoring Criteria	
Section	Requirement
Executive Summary (10 points)	Contains objective statement
	Brief description of problem to be solved (purpose of design)
	Discussed the planned approach to achieve all objectives
	Summarizes main point from subsequent sections
	Clear and concise, uses proper grammar; 2 pages max.
Management Summary (5 points)	Overall description of team organization (leadership, sub teams and responsibilities)
	Schedule includes key actions through report deliverable and presentation
	Schedule includes detail on design, prototype and testing phases
	Schedule includes detail for subcomponent design
Design Trade Studies (5 points)	Mission requirements decomposed into aircraft subsystem requirements
	Sensitivity study of design parameters discussed, major design drivers detailed
	Review of configurations considered
	Describe concept weighting, selection process and results (e.g. configuration, motors, props, etc.)
Technical Innovations (10 points)	Detail any unique design considerations or technologies used by the team
	Describe "mission model" used for predicting system performance
	Mission model description includes equations
	Mission model discusses source of inputs (aero, propulsion, environment, etc.)
	Mission model includes discussion of assumptions and uncertainties
Design Definition (15 points)	Sub system discusses all key components (airframe, propulsion system, electronics)
	Structural analysis of key structural components, includes max. expected loads
	Document dimensional parameters of final design
	Document mission performance for final design
	Weight and balance of final design
	Provide estimate of aircraft lift and drag and method of prediction
	Provide estimate of static and dynamic stability and method of prediction
Drawing Package (15 points)	Drawing of aircraft: front view
	Drawing of aircraft: side view
	Drawing of aircraft: top view

	Drawing of aircraft: isometric view
	Structural arrangement drawing present (showing mechanical interfaces, spars, ribs, gear, etc.)
	Systems layout/location drawing present (showing motor, servos, speed controllers, batteries, receiver, etc.)
	Quality, detail and thoroughness of drawings
Fabrication Methods (10 points)	Manufacturing processes investigated, discussed, and compared
	Discussion on how investigated materials and methods were down-selected
	Final manufacturing process presented in detail
Test Plan (5 points)	Discussion of major tests planned (i.e. wind tunnel, structural, propulsive, flight, etc.)
	Test objectives for each
	Describes proposed set up and data to be collected for each
Flight Test Results (5 points)	Describe the demonstrated performance of key subsystems (propulsion, structure, electrical, etc.)
	Compare to predictions and explain any differences and improvements made
	Describe the demonstrated performance of your complete aircraft solution (flight testing)
Presentation and Organization (20 points)	Proper grammar, spelling, formatting
	Figures & texts taken from published works are referenced
	Reference list at the end of the document in numerical order as cited in the text
	Logical progression of report; easy to read with headings, etc.
Total (100 points)	

3.3: Presentation Scoring

Table 3: Final Presentation Scoring Criteria	
Section	Requirement
Team Organization (10 points)	Showed breakdown of team roles and responsibilities throughout competition
	Gantt chart discussing timelines proposed vs. actual
Originality (15 points)	Major subassemblies not off-the-shelf (e.g., airframe — excludes motors, props, batteries)
	Innovative engineering used to solve the competition challenges (out of the box ideas)
Engineering (25 points)	Presentation of concepts considered and down selection process to final design concept
	Validation/testing process (bench testing, wind tunnel, flight testing, etc.)
	flight video
Drawings (15 points)	Aircraft 3D CAD renderings of vehicle and pertinent sub-systems
	Structural arrangement drawing or rendering (showing mechanical interfaces, spars, ribs, gear, etc.)

Publicity (15 points)	Use of graphics or charts to convey vehicle performance
	Successfully conveyed why final design was the rational choice given their assumptions
Presentation (15 points)	Clear, easy to follow, logical presentation of material
	Good speaking presence and breakdown of material across presenting team members
Public Relations (5 points)	Successfully fielded audience and/or judges' questions
Total: 100	

3.4: Aircraft Mission Performance Scoring

- Safety Checks (Pass/Fail)
 - Bench Check Completion
 - Hover Test Completion

3.4.1: Flight Performance Course Scoring

Points awarded for the flight performance course will be broken down into interval achievements towards mission completion. The scoring breakdown consists of:

1. The aircraft rising vertically out of the start/end VTOL zone (Pass/Fail)
 - a. Failing this portion results in zero points for the course attempt.
2. Properly maneuvering around the course as described in Section 2.3.1 with controlled VTOL touch downs inside each of the designated landing zones (Pass/Fail)
 - a. Failing this portion results in zero points for the course attempt.
3. Controlled final landing inside the start/end VTOL zone (Pass/Fail)
 - a. Failing this portion results in zero points for the course attempt.
4. Course time score (points awarded as shown in Table 4)
 - a. Time for the first three laps will be recorded for this scoring section.
5. Course lap score (points awarded as shown in Table 5)
 - b. Total number of laps completed will be recorded for this scoring section.
6. Course payload fraction score (points awarded as shown in Table 6)
 - a. Payload fraction defined as (payload weight)/(MTOW)

Table 4. Flight Performance Course Scoring

Flight Performance Course Time Score (50 possible points); N = Total Number of Teams	
n th Place (n = 1 for team with fastest time)	$50 * ((N + 2) - n) / (N + 1)$

Table 5. Flight Performance Course Lap Scoring

Flight Performance Course Lap Score (50 possible points); N = Total Number of Teams	
n th Place (n = 1 for team with most laps)	$50 * ((N + 2) - n) / (N + 1)$

Table 6. Flight Performance Course Payload Fraction Scoring

Flight Performance Course Payload Score (50 possible points); N = Total Number of Teams	
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n^{th} Place ($n = 1$ for highest payload fraction)	$50 * ((N + 2) - n) / (N + 1)$
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3.4.2: Autonomy Course Scoring

Each team to successfully complete the autonomy course will be awarded full points.

Table 7: Autonomy Course Scoring

Autonomy Course Score (150 possible points)	
Successfully Completed Course	150

4.0: Aircraft Design Restrictions and Requirements

1. Maximum Weight: Aircraft maximum take-off weight must be no more than 15 lb.

- This weight limit includes any payload used for the flight performance course. For example, a 13-lb aircraft would fly at the maximum allowable 15 lb when carrying a 2-lb payload.
- **A vehicle will not be allowed to fly if it is over 15 lb. This means that the aircraft, all systems, power sources, and any payloads are all counted against this 15-lb maximum.**

2. Power System: The aircraft must be electric (electric power source and electric motors).

3. Propulsion System Batteries: For the purpose of competition safety, unaltered commercially available LiPo batteries of six cells or less (6S or less) may be used. The team needs to select the proper capacity for their vehicle.

- Each battery is limited to 100 W-hr.
- Each battery used in the propulsion system must have a fuse directly in-line with its positive terminal that has a maximum continuous current rating equal to or less than the maximum continuous discharge rating of the battery. i.e. A-hr x maximum continuous (constant) Coulomb rating.
- Multiple batteries following the above requirements can be combined in parallel for the propulsion system.
 - Each battery must be identical and have its own fuse on its positive lead.
 - Note* A different capacity battery can be used to power the ESC/flight control system that follows the battery guidelines specified throughout the RFP.
- All LiPo batteries at the competition must be charged and stored inside a LiPo charging bag. They are only allowed out while being actively used with the aircraft.
- Onsite charging is allowed and power will be provided by ARL.
- A 6S LiPo battery has a nominal voltage of 22.2V (3.7V/cell) and charging these batteries to the manufacturer's specified capacity is allowed (typically 4.2V/cell).

4. Power Source: Aircraft must have a separate power source for the flight control system. A LiPo battery that also follows the specifications under point 3 and throughout the RFP must be used, except that it can differ in capacity from the propulsion system battery/batteries.

5. Minimum Payload: The aircraft must be capable of carrying a minimum payload of 2 lb (907 g) for use in the flight performance course. There is no maximum payload limit, as long as teams stay within the 15-lb MTOW limit.

6. Maximum Dimension: 6.5 ft (198 cm), including the propellers/rotors in their most outstretched positions. i.e. vehicle must fit within a 6.5 ft diameter cube.

7. VTOL: The aircraft must take off and land vertically.

8. Kill Switch: The remote controller must demonstrate a lost-link power-cut functionality.

9. Shunt Plug: The purpose of the shunt plug is to provide an easy and quick way to manually disarm the aircraft.

- A shunt plug must be wired between the leads of the battery system and the electronic speed controller for manual disarming and arming of the aircraft's power system.
- The shunt plug must be red.
- The shunt plug must be removable with only one hand and without any tool.
- **The tip of the shunt plug, where someone would grab it, must be located outside the dotted line as shown in Figure 6, below.**
- The dotted line, if extended both into and out of the page, creates a box around the aircraft that extends outward from the rotors by 6 inches in all directions. Hint: Placing the shunt plug aft of the vehicle may be the best solution for aerodynamic and stability purposes in forward flight.
- A physical switch mounted on the drone would not be permitted and is not considered a valid shunt plug.

10. Transmission Frequencies: Must follow US Federal Communications Commission (FCC) Part 15 rules for transmission frequencies and International Telecommunication Union (ITU) Region 2 frequency allocations. This means that telemetry, video, and control transmitters must operate on 902-928 MHz, ~2.4 GHz or ~5.8 GHz.

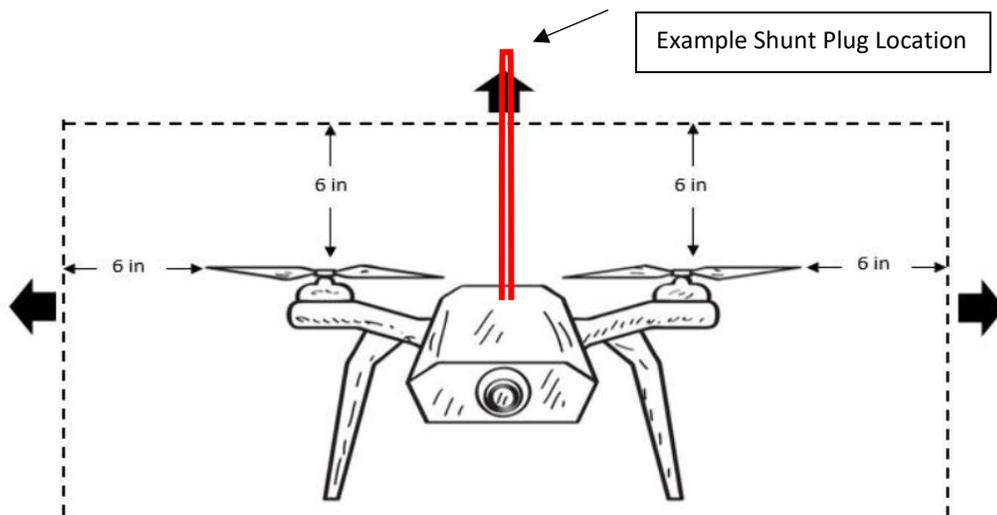


Figure 6. Allowable Location for Propulsion System Shunt Plug

5.0: Prizes

Competition awards will include 1st through 3rd place overall trophies and the following award amounts:

- 1st – \$2000
- 2nd – \$1000
- 3rd – \$500

A plaque will be awarded to first place winner of the combined final technical report and presentation scoring. The second and third place teams in this scoring combination will receive certificates.

Monetary prizes will also be awarded following the review and scoring of the preliminary design reports for the 1st through 3rd best scores as follows:

- 1st – \$750
- 2nd – \$500
- 3rd – \$250

Additionally, awards may also be given for various “judges choice” categories to be announced at the end of the competition.

6.0 Expenses and Support

VFS will not provide any travel or accommodation support for competing teams, or pay or reimburse any other expenses. Teams are encouraged to search for university or company sponsors for travel, accommodation, equipment, etc., and are free to display any sponsors logos on their team shirts and on their aircraft.

7.0 Pilot Requirements

Team pilots must hold an FAA Part 107 Drone Certification in order to fly at the ARL facility and may not be a US federal government employee. Below is a summary of details:

- Pilot(s) must be identified at the time of the **March 1st** when the fly-off team list is due to ARL.
- Teams can name more than one (1) pilot as long as all pilots hold FAA Part 107 Drone Certification and be within the five (5) team member limit for the fly-off.
- **Pilot FAA Part 107 Certification and UAS Pilot Certifications & Training/Experience list required by April 25 to ARL**
 - Provide an estimate of cumulative UAS flight (pilot) time
 - Submit a list of any other UAS certifications, including FAA Part 107 certification, held and training taken, along with date achieved and name of administering organization for each
- The pilot will be required to complete ARL pilot test assessment in-person and in the presence of an ARL employee on June 1 to include the following
 - Pilots of US-based teams must hold an FAA Part 107 certification.
 - Pilots of non-US-based team may hold an FAA Part 107 certification or hold a UAS pilot certification from a certifying organization
 - Perform the ARL-administered pilot-in-command checklist on site at Graces Quarters – checklist provided by ARL as part of the USA Pilot Certification & Training document

8.0 ARL Component List Requirements and Other Documentation

In order to compete in the fly-off at the ARL government facility the following documentation will be required by the competing teams which will be due in February (exact date TBD). **Note: No US Department of Defense funds shall be used for procurement of components used in the air or on the ground.**

It is expected that each team will fly a single UAS configuration in the competition. Therefore, each team is expected to submit a single component request form that addresses alternative components being considered in their final UAS design.

Refer to Attachment 1 for ARL UAS Component Information worksheet, which includes an example diagram. Below is information to help complete the worksheet.

- Once the component request form is submitted, no changes are permitted.
 - Motors, rotors (propellers), batteries, and airframes are not required to be on the component request form, and as such are allowed to change after its submission. The only limitations on these components are related to size and battery constraints that are specified in the RFP.
- The vehicle maximum dimension and maximum take-off weight must adhere to the specifications in Section 4 of the RFP.
- UAS configurations will be inspected to ensure each component complies with the make/model that was stated in the UAS component information request form.
- Any device that transmits and/or receives signal must be included in the component request form. “Device” is defined as a component in each row of Section II of the component request form. For example, if a team is considering alternative flight controllers/autopilots, data radios, or payloads (i.e., sensors), then include all required details for each alternative “device” (aka “component”) – up to 3 – being considered.
 - LiDAR devices must be included in the component request form. For example, if a team is considering a UAS configuration where the payload (for purposes of the ARL documentation, not the flight performance course payload) is one LiDAR, one EO camera, one IR camera, one altitude sensor, one GPS, and one depth camera, this is acceptable. However, all required details of all of these sensors must be included in the component request form. If a team is considering multiple (up to 3) alternatives for any of the aforementioned example sensors, then the details for all alternatives must be included in the component request form.
- Encryption is optional. If encryption is used, please provide all requested information on the network diagram and UAS component sheet.

9.0 Autonomy

This RFP has an *Appendix A, “Achieving Autonomy: An Overview,”* which is meant to serve as a helpful guide on the autonomous flight portion of the competition.

10.0 Questions

Questions should be sent to fly@HQ.vtol.org. The Frequently Asked Questions (FAQ) document will be posted to the competition site on a semi-monthly basis, starting the end of September 2022.

- Questions should be submitted **by the 15th of every month**

- Updated FAQ document will be posted to the DBVF on site the 30th of every month. vtol.org/awards-and-contests/design-build-vertical-flight

11.0 Disclaimers

Safety is paramount in this competition. The rules are designed specifically to minimize risk to all participants and to comply with US government restrictions at the test site. VFS assumes no responsibility for any actions caused by any participants of the DBVF Competition.

These rules are subject to change. Any updates will be published at the end of a revised document and it will be posted to **www.vtol.org/FLY**. All registered competitors will be notified of any clarifications on the rules or necessary adjustments.

Participation in the competition explicitly gives permission to VFS to use text, graphics, photographs, and video documentation of the competition and all competitors for educational and promotional purposes only. The competition event, participating schools, and team names may be the subject of VFS *Vertiflite* magazine articles, web page postings, social media, or other forms of publicity.

12.0 Final Word

VFS wishes you all the success possible in undertaking the second annual VFS Design-Build-Vertical Flight Competition and we look forward to meeting you at the fly-off competition. Good luck!